

Appl. No. : 10/009,281
Filed : April 19, 2002

AMENDMENTS TO THE SPECIFICATION

Page 1, please amend the paragraph beginning on line 7 as follows:

The release mechanism according to the invention is developed for use in missiles, and in particular, but not exclusively, in rocket accelerated penetrators. Rocket accelerated penetrators are often kept in their storing and standby state with the main parts thereof not assembled. This means that the part having control fins, the fin cone, and the rocket motor proper is assembled to the penetrator at the moment before the missile is launched from the launcher. The penetrator, which is in form of an arrow like body having substantial mass, is lying in standby position in a translation tube within the rocket motor and with the pointed end thereof supported in the control fin part. How the assembly operation happens is more detailed described in the ~~priority founding Norwegian~~ U.S. patent application no. 19992739 6,659,393 issued December 9, 2003.

Page 3, please amend the paragraph beginning on line 28 as follows:

The propulsion means for translation of the projectile through the translation tube within the rocket motor is described in closer detail in ~~depending Norwegian~~ U.S. patent application no. 19995142 6,647,889 issued November 18, 2003. The mechanism for translation of the projectile and subsequent locking to the rocket motor is described in closer detail in ~~depending Norwegian~~ U.S. patent application no. 19995141 6,640,720 issued November 4, 2003.

Page 3, please amend the paragraph beginning on line 34 as continuing on page 4 as follows:

Fig. 3 shows the rear end of the penetrator 1 when the penetrator is translated through the control fin part 5. The rear end of the penetrator 1 interlocks to the control fin part 5 after this translation. How this happen is described in closer detail in ~~Norwegian~~ U.S. patent application no. 19992739 6,659,393 issued December 9, 2003.